

GAlileo Survey of Transient Objects Network (GASTON) project

Searching for dark matter using the Galileo satellites

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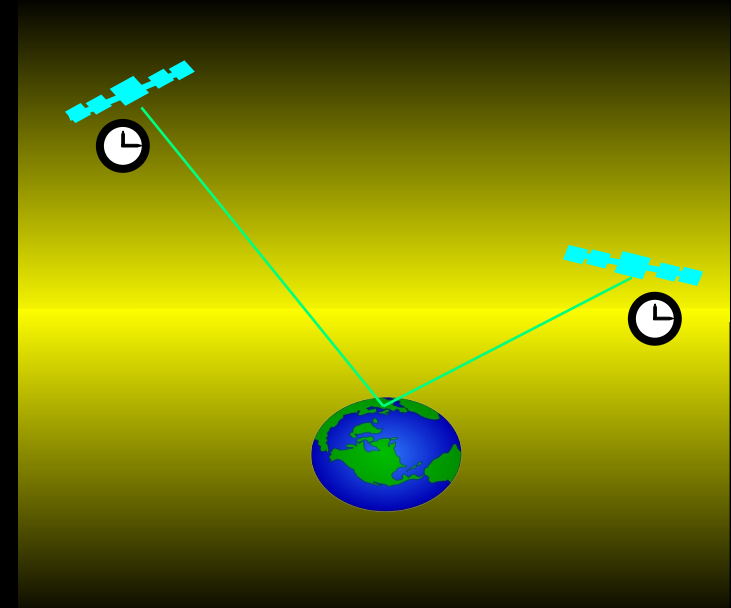
ESA, Centre Spatial de Toulouse

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8th International Colloquium on Scientific and Fundamental Aspects of GNSS
September, 15th

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Fundamental constants are free parameters inherent to the theory that introduces them.

Why a constant could vary ?

Signature of new physics beyond the Standard Model

=> Dimensionless combinations of constants are measured:
fundamental parameters

Damour T., J. Donoghue, 2010, Phys. Rev. D 82
Stadnik Y., V. Flambaum, 2015, Phys. Rev. Let. 115
Arvanitaki A., et al, 2015, Phys. Rev. D 91
Hees A. et al, 2018, Phys.Rev.D 98

Standard Model:
22 constants

Rest mass of particles
Proton, electron, quarks...

m_p m_e m_q

Planck constant \hbar

Elementary charge e

Speed of light c

...



Use atomic clocks onboard Galileo satellites to **test transient variations of fundamental parameters**

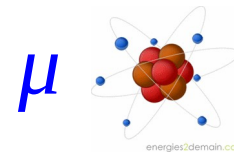
The fine structure constant



Strength of the electromagnetic interaction

$$\alpha = \frac{1}{4\pi\epsilon_0} \frac{e^2}{\hbar c} \approx \frac{1}{137}$$

Proton-to-electron mass ratio



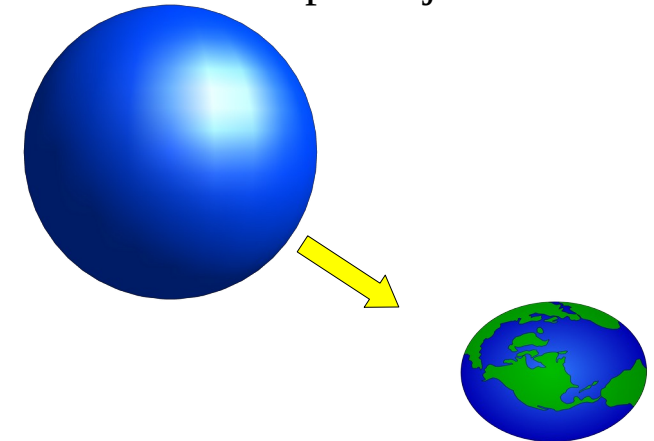
$$\mu = \frac{m_p}{m_e} \approx 2000$$



Dark matter (DM) models as a **test bench** of our method

- Recent investigation: DM could be on the form of clusters or macroscopic structures.
- Such structures could cross regularly the Earth!
=> **DM transients**

DM macroscopic object





Apparent spacetime variation of fundamental constants

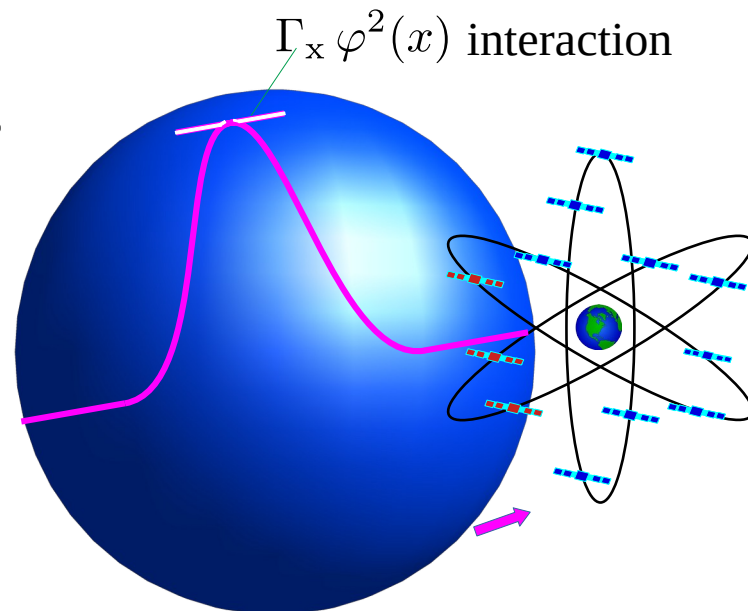
Effective fermion mass $m_f^{\text{eff}}(x) = m_f^0 (1 + \Gamma_f \varphi^2(x))$

Effective fine structure constant $\alpha_{\text{eff}}(x) = \alpha_0 (1 + \Gamma_\alpha \varphi^2(x))$

Γ_x characterises the strength of the coupling between the DM field and electromagnetism/fermions

$$\Gamma_x \equiv \Gamma_\alpha \text{ or } \Gamma_f$$

$\varphi(x)$ dark matter field





Galileo: a giant detector for new physics

Shift in energy levels inside the transient

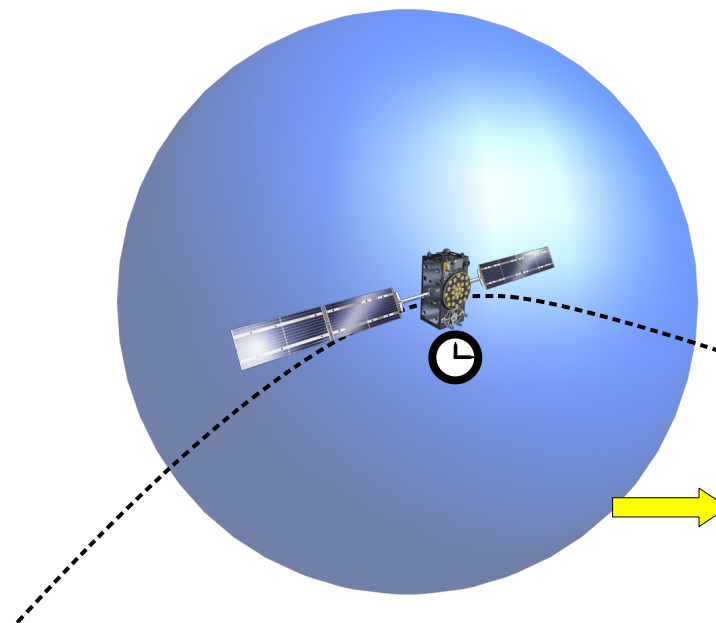


Transient shift in atomic clock frequencies

$$\frac{\omega(t) - \omega_0}{\omega_0} = \kappa_\alpha \frac{\Delta\alpha(t)}{\alpha_0} + \sum_f \kappa_f \frac{\Delta m_f(t)}{m_0}$$

Sensitivity coefficients

A. Derevianko & M. Pospelov, Nature Phys., vol.10, 2014
 M. Pospelov et al., Phys. Rev. Lett. 110, 2013
 L. Visinelli & J. Redondo, arXiv:1808.01879,2018
 A. Banerjee et al., arXiv:1902.08212, 2019





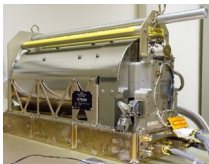
Gaston project: large extension of

B. M. Roberts et al., Nature Com., vol.8, 2017

More stable clocks



Passive H-maser clocks
onboard Galileo satellites



Finer analysis of
systematic effects



Intensive SLR campaign
over 3 months



More complex
theoretical model

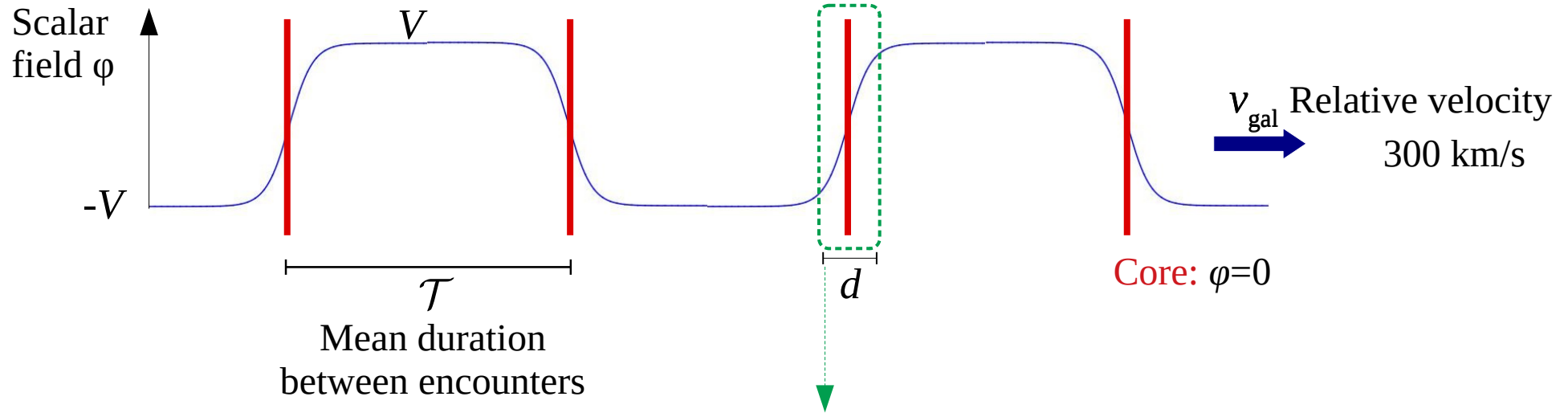


Extension to *large* transient
objects: well beyond 10^3 km

Image maser: ESA



Gaston model of energy density for a network of transient objects



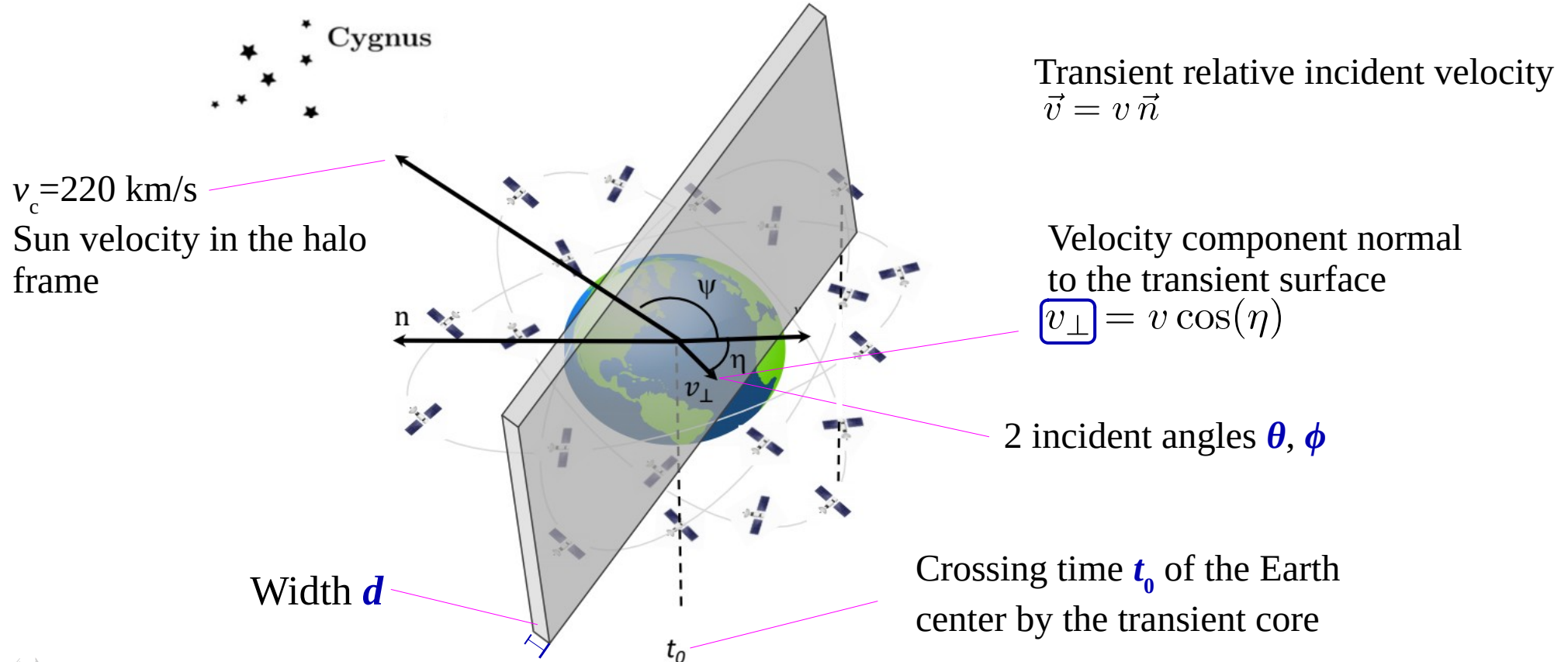
Local dark matter energy density ρ_{DM} :
upper limit for the energy density

Domain wall solution

$$\varphi(z) = V \tanh\left(\frac{z}{d}\right)$$

$$\frac{V^2}{d} \leq \frac{3}{4} v_{\text{gal}} \tau \rho_{\text{DM}}$$

Our modelling depends on a set ξ of 5 parameters: $\xi \equiv (d, v_{\perp}, \theta, \phi, t_0)$

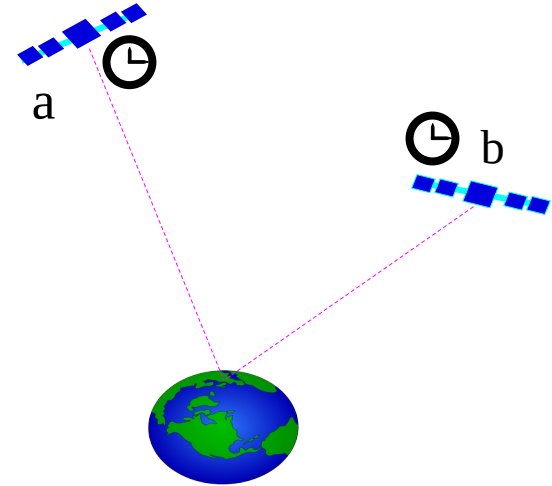




$s_{ab}(t)$ Clock bias between two clocks 'a' and 'b'

With a time sample $\Delta T = 30s$

$$s_{ab}^{(1)}(t) = \frac{s_{ab}(t) - s_{ab}(t - \Delta T)}{\Delta T}$$



For each pair of Galileo clocks

- Moving windows of 14h in the $s_{ab}^{(1)}(t)$ series
- Remove a linear regression in each window



A bank of signal templates $\bar{s}_{ab}(\xi, t)$ was modeled

Modeled signal

$$s_{ab}^{(1)} = h \frac{\bar{s}_{ab}(t, \xi)}{\Delta T}$$

Model dependent amplitude h

$$h = \underbrace{\Gamma_x}_{\text{Parameter to constrain}} \frac{3}{2} \frac{R_{\text{sat}}}{\omega_{\text{sat}}} \underbrace{\rho_{\text{DM}} v_{\text{gal}} \mathcal{T}}_{\text{DM model}}$$

Parameter to constrain

DM model

$$s_{ab}^{(1)} = h \bar{s}_{ab}(t; \underbrace{d, v_{\perp}, \theta, \phi, t_0}_{\text{Event parameters}}; \underbrace{\vec{r}_a(t), \vec{r}_b(t)}_{\text{Galileo satellite orbital data}})$$

Event parameters

Galileo satellite orbital data

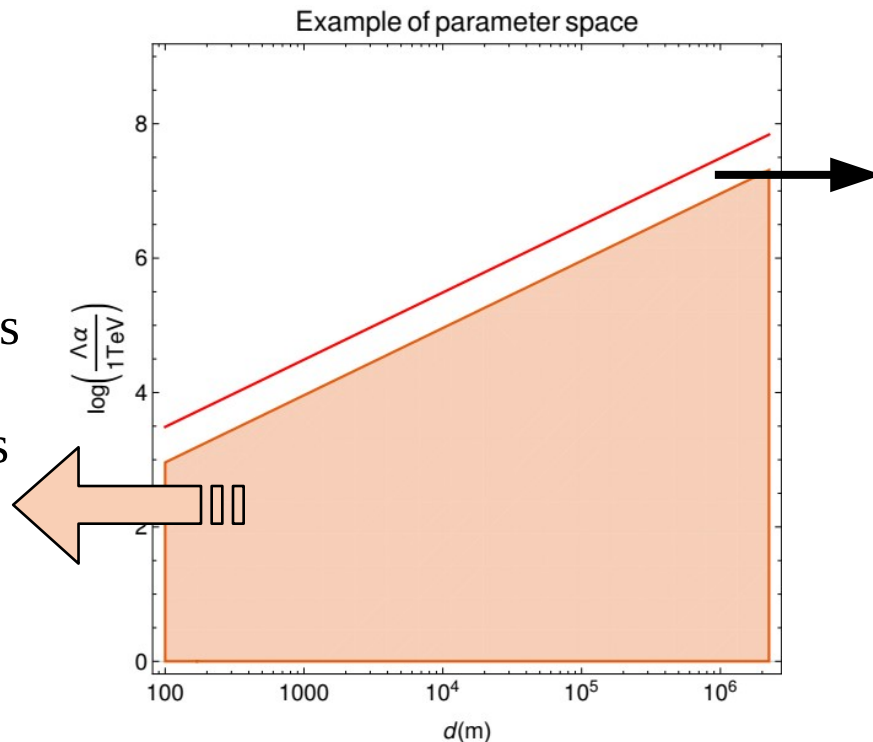
$$\xi \equiv (d, v_{\perp}, \theta, \phi, t_0)$$



Level 1 : Maximum reach analysis

Look for frequency spikes
in clock differences
between pairs of satellites

Goal : Excluding regions
of the parameter space



Level 2:
Frequentist approach

Correlation analysis over
the whole set of available
Galileo satellites

Goal : Detection of
possible events



Physical model for transient events

$$h(\Gamma_x) \bar{s}_{ab}(t; d, v_{\perp}, \theta, \phi, t_0; \vec{r}_a(t), \vec{r}_b(t))$$

↓
Parameter to
constrain

↓
Physical pattern

Experimental data series

$$s_{ab}^{(1)}(t)$$

Theoretical lower reach for event configuration

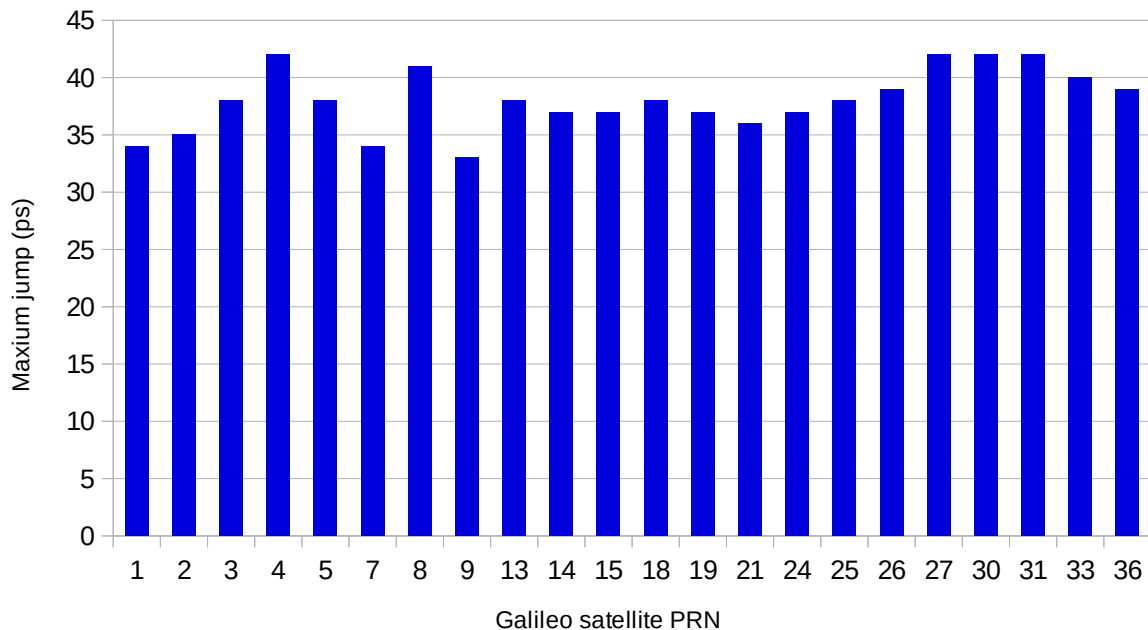
$$s_*(d) \leq \bar{s}_{ab}(t; d; v_{\perp}, \theta, \phi, t_0; \vec{r}_a(t), \vec{r}_b(t))$$

Choosing the *worst* **transient** and **satellite**
configuration at 95% level



Experimental data series

Considering for each satellite the maximum frequency spike with respect to all other satellite clocks



$$s_{ab}^{(1)}(t)$$

Largest Observed effect

$$s_{a,\max}^{(1)} = \max_{b \neq a} \max_{t_i} s_{ab}^{(1)}(t_i)$$

Considering the statistical clock noise:

$$\Delta T \min_a s_{a,\max}^{(1)} = 42\text{ps}$$



Physical model for transient events

$$s_{ab}^{(1)} = h \frac{\bar{s}_{ab}(t, \xi)}{\Delta T}$$

Experimental data series

Largest Observed effect

$$s_{a,\max}^{(1)} = \max_{b \neq a} \max_{t_i} s_{ab}^{(1)}(t_i)$$

Theoretical lower reach for event configuration

$$s_*(d) \leq \bar{s}_{ab}(t; d; v_{\perp}, \theta, \phi, t_0; \vec{r}_a(t), \vec{r}_b(t))$$

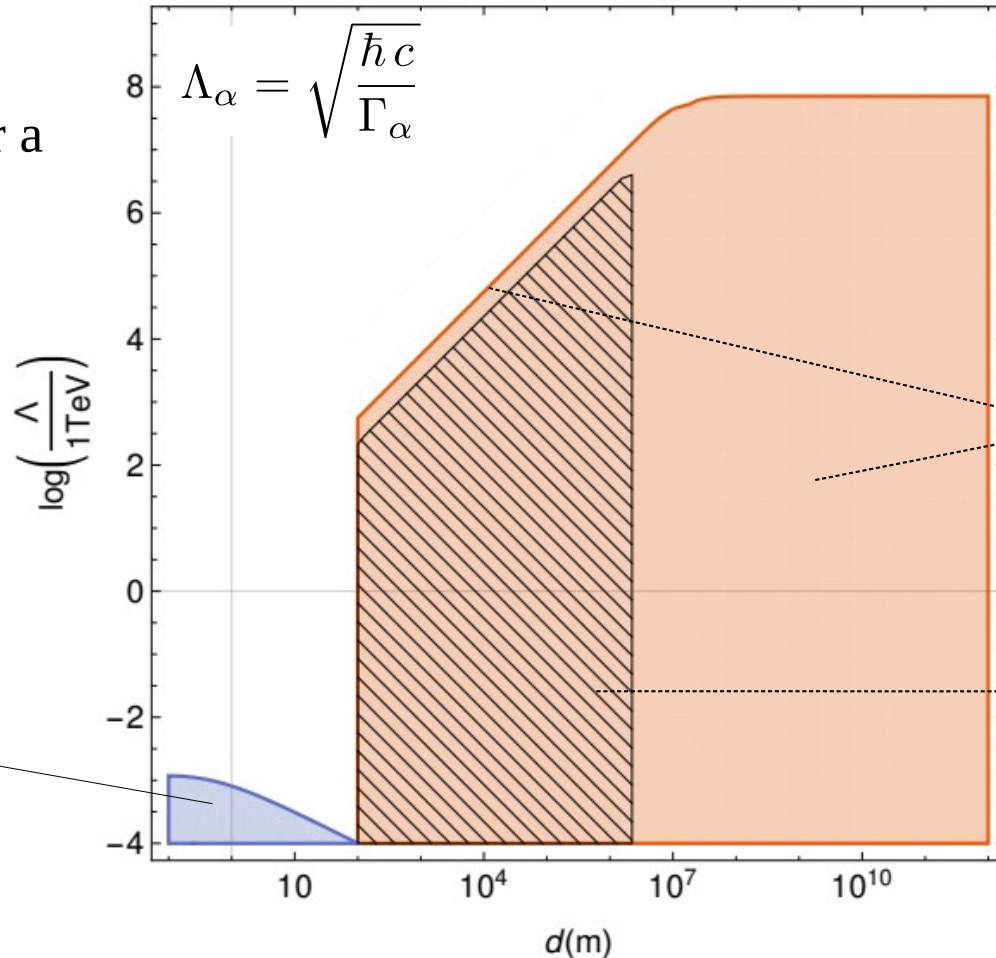
Constraint

$$h(\Gamma_x) s_*(d) \leq 42\text{ps}$$



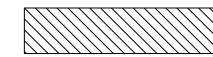
Size d of the transient restricted to 10^{12} m for a value of $\mathcal{T} = 1$ months

Constraints on the parameter space for $\mathcal{T} = 1$ month



$$h(\Gamma_x) s_*(d) \leq 42\text{ps}$$

Gaston constraints by maximum reach analysis



Hatched region:

B. M. Roberts et al.,
Nature Com., vol.8, 2017

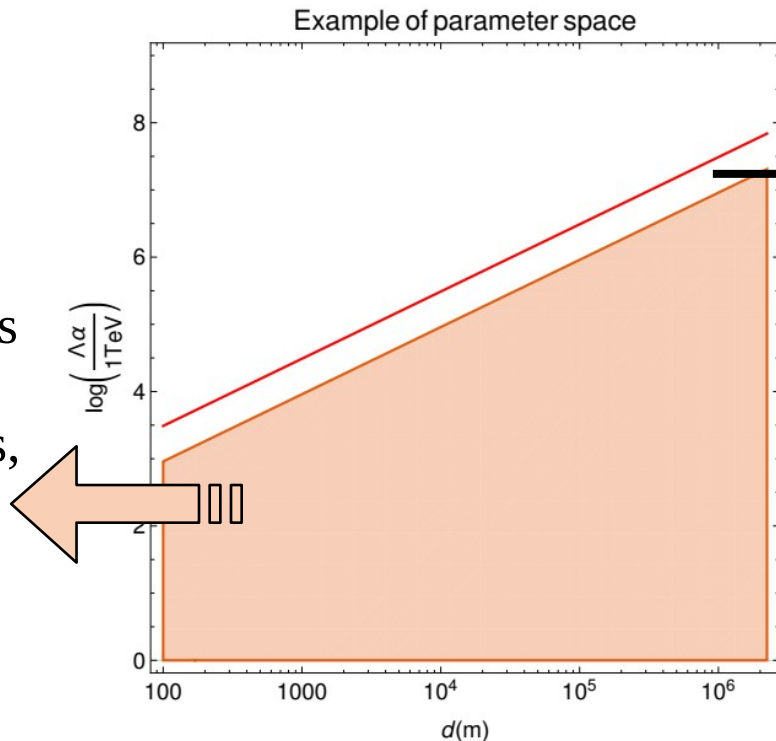
Signal propagation

B. Bertrand and P. Defraigne,
Adv. in Space Res., Vol. 66,
Issue 12, 2020



Level 1 :
Maximum reach analysis

Look for frequency spikes
in clock differences
between pairs of satellites,
over the 3 months
campaign



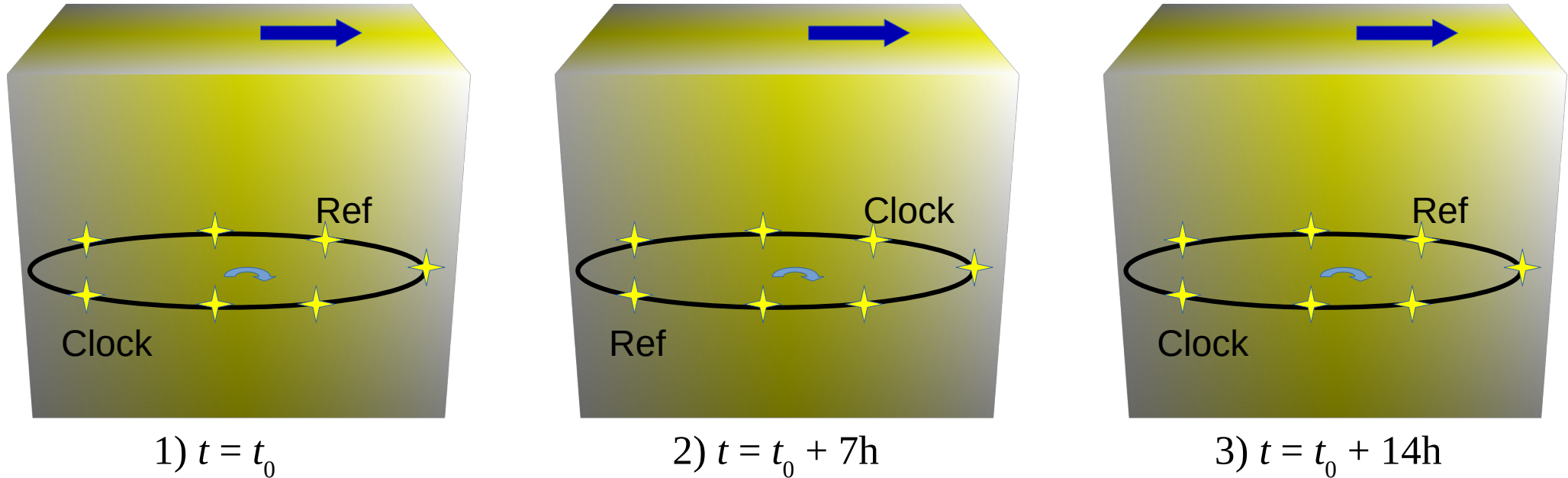
Level 2:
Frequentist approach

Correlation analysis over
the whole set of available
Galileo satellites

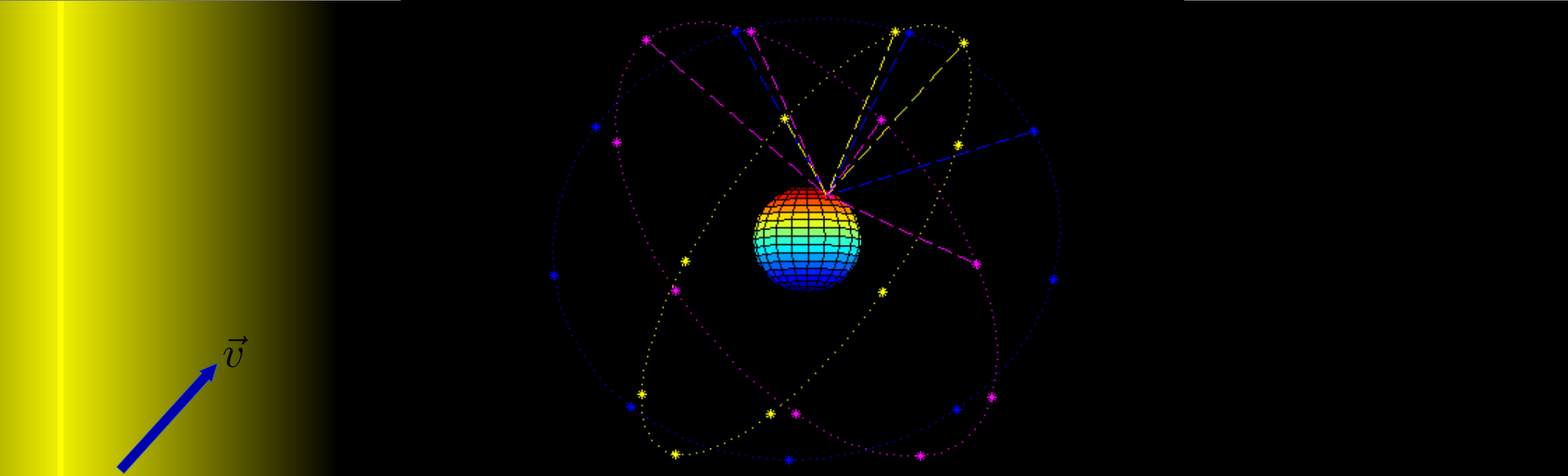
Goal : Detection of
possible events



Gradient of the scalar field \rightarrow Phase difference w.r.t. the reference clock



Periodic signature (orbital period) in the comparison between the clock and the reference clock



$s_{ab}^{(1)}(t) \cdot 10^{-8}$

Relative
frequency

- E03-E19 — E07-E09
- E04-E08 — E30-E01
- E14-E18 — E33-E15
- E27-E31 — E21-E24
- E05-E26
- E13-E25

Global animation:
Bruno Bertrand

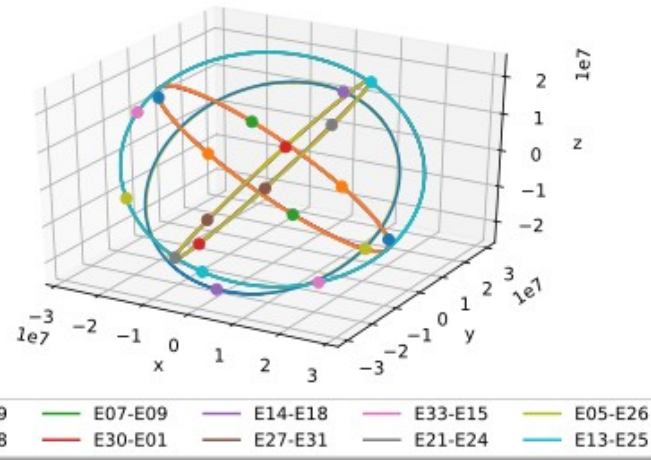
Galileo satellites:
Lukas Rohr, CC BY-SA 3.0

Time [h]



$$\mathcal{L}(D|S) \propto \exp\left(-\frac{1}{2} \underbrace{[d - h\bar{s}(\xi)]^T}_{\text{Data series}} \underbrace{\mathbf{C}^{-1}}_{\text{Covariance matrix}} [d - h\bar{s}(\xi)]\right)$$

Independent pairs of clocks a,b:
diagonalises \mathbf{C}





Gaussian likelihood

$$\mathcal{L}(D|S) \propto \exp\left(-\frac{1}{2} [d - h\bar{s}(\xi)]^T \mathbf{C}^{-1} [d - h\bar{s}(\xi)]\right)$$



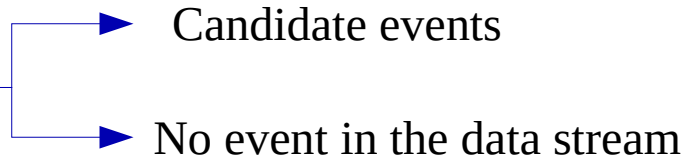
Template-dependent signal-to-noise ratio (SNR) ρ_ξ



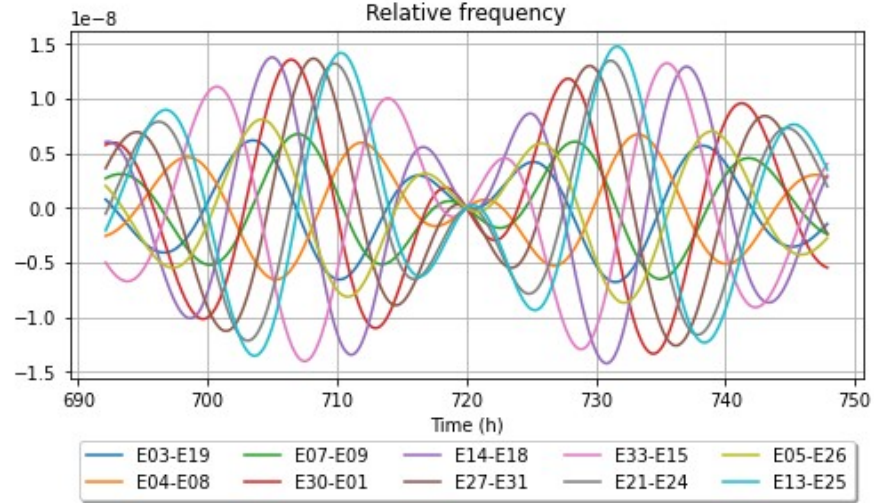
$$\rho_\xi = \frac{h_\xi}{\sigma_{h_\xi}} = \frac{d^T \cdot C^{-1} \cdot \bar{s}_\xi}{\sqrt{s_\xi^T \cdot C^{-1} \cdot \bar{s}_\xi}}$$

Detection threshold ρ_{thres} with N_t templates from max-SNR distribution

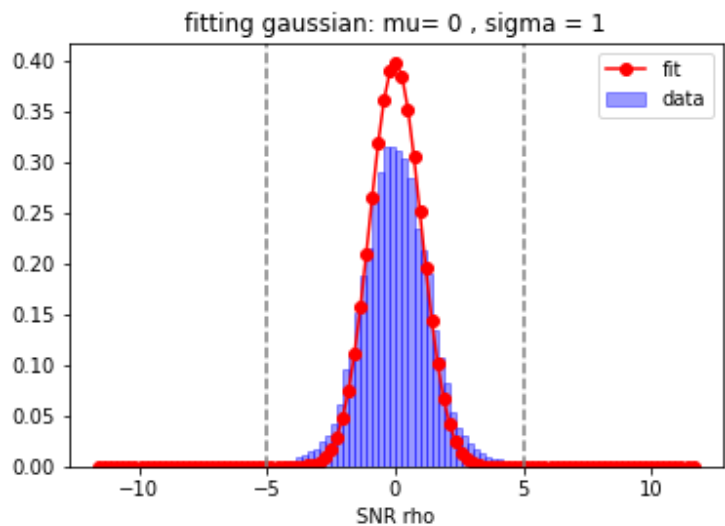
$$\rho_{\text{thres}} = \sqrt{2} \operatorname{erf}^{-1}\left(0.95^{\frac{1}{N_t}}\right)$$



Modeled signal $s_{ab}^{(1)} = h \frac{\bar{s}_{ab}(t, \xi)}{\Delta T}$

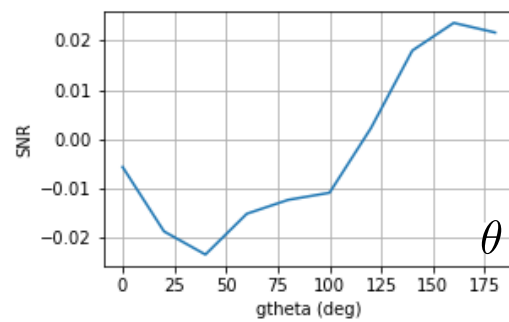
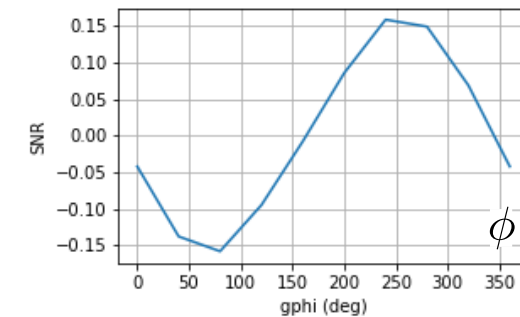
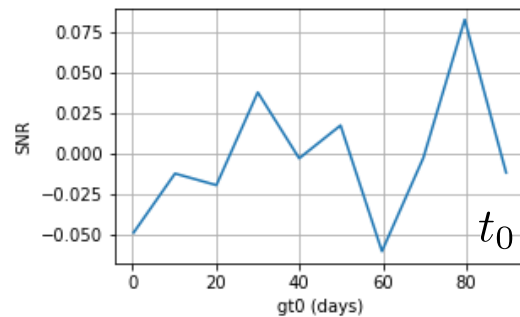
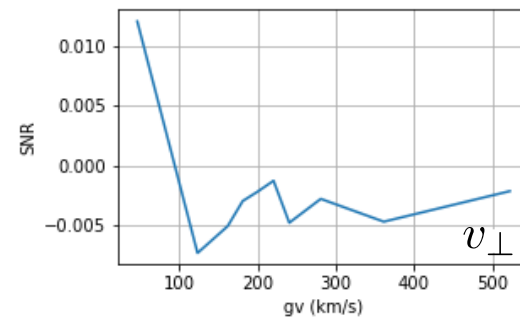
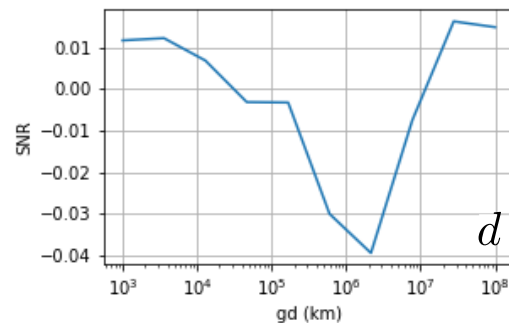


$d = 10^7 \text{ km}$
 $t_0 = 720 \text{ h}$
 $v_\perp = 300 \text{ km/s}$
 $\theta = 45^\circ \quad \phi = 45^\circ$



$$\rho_{\text{thres}} = 5.02$$

Over 1 year of observation, up to 6% of the events with $\text{SNR} > \rho_{\text{thres}}$ according to the observation epoch.

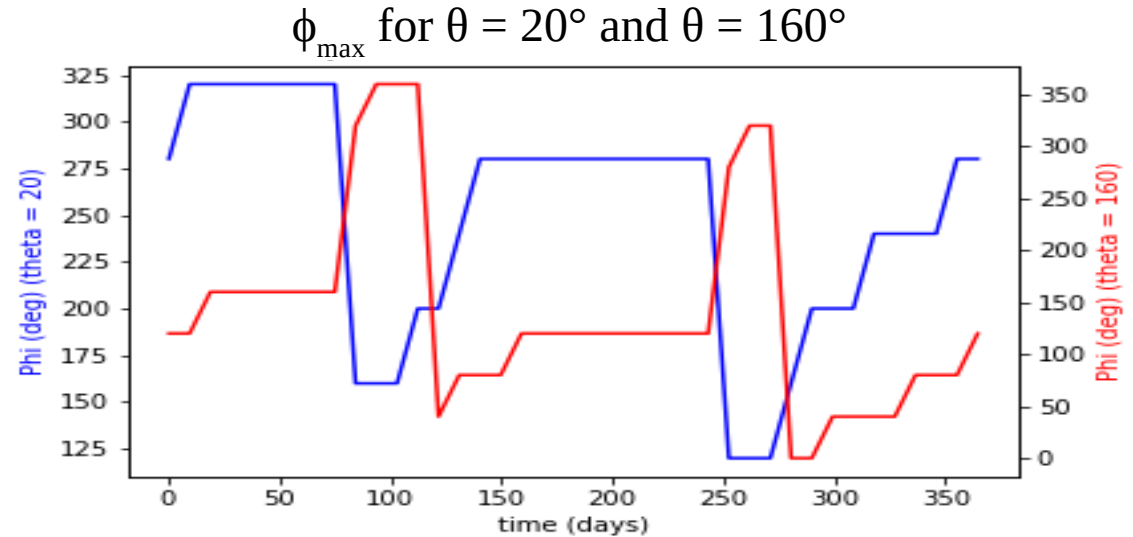
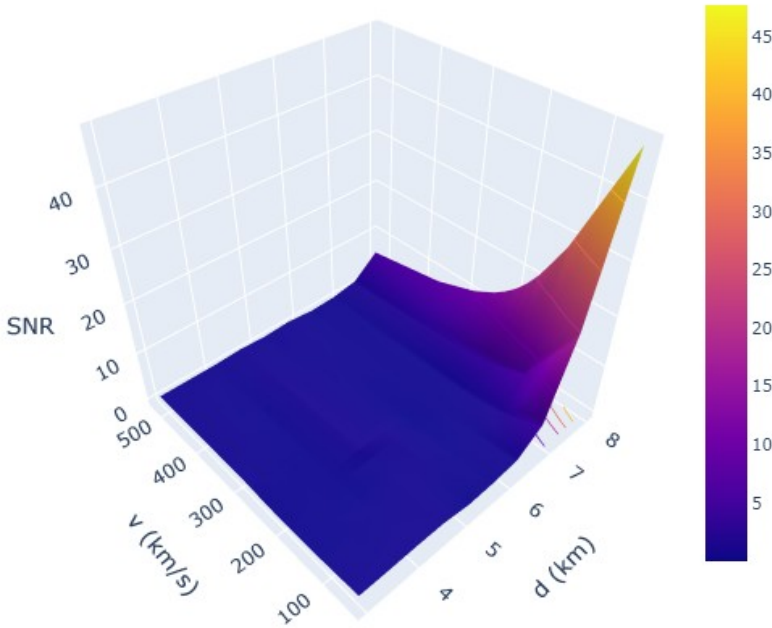


The SNR varies with the incidence direction (ϕ, θ)



The SNR increases for large transients ($d \sim 10^8$ km) and low velocities ($v_{\perp} < 200$ km/s)

- Possible correlations with orbital errors (SRP mismodelling?)
- Maxima of SNR: The incident direction of the transient changes with its time of arrival (~ 150 days period)





Transients DM objects

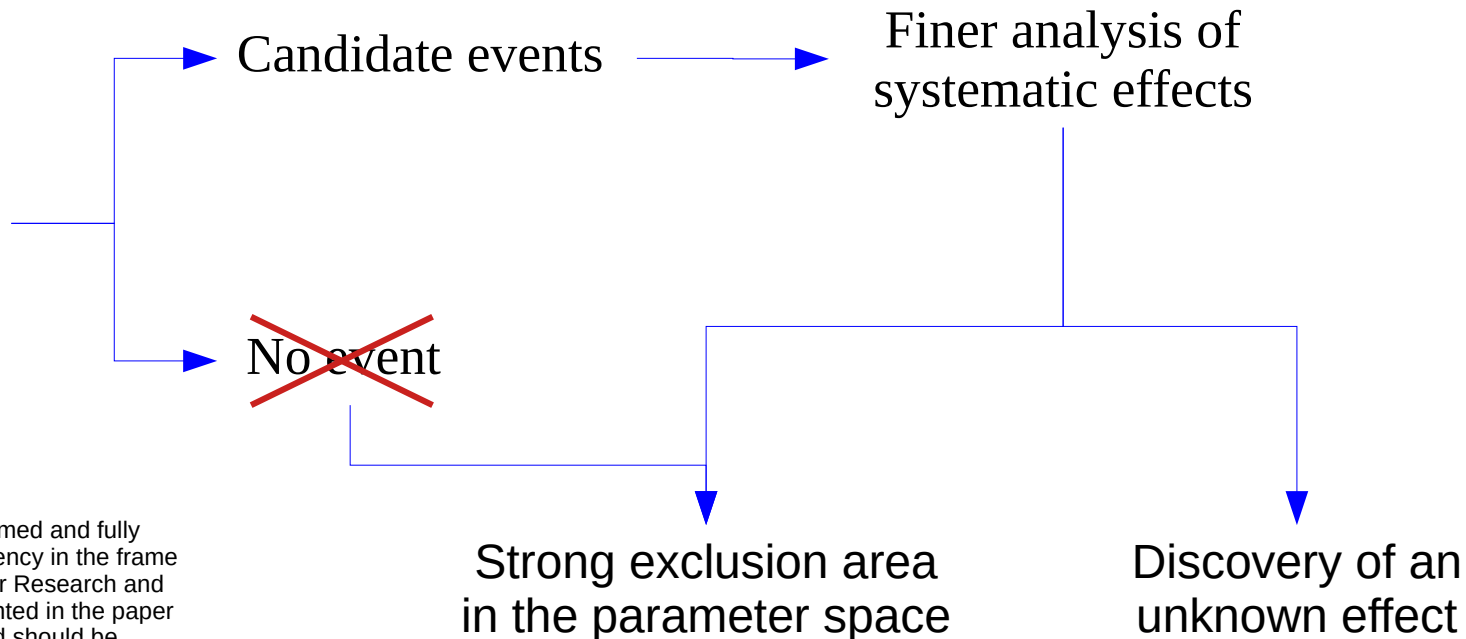


Full time correlation over the 21 satellites



Intensive SLR campaign over 3 months

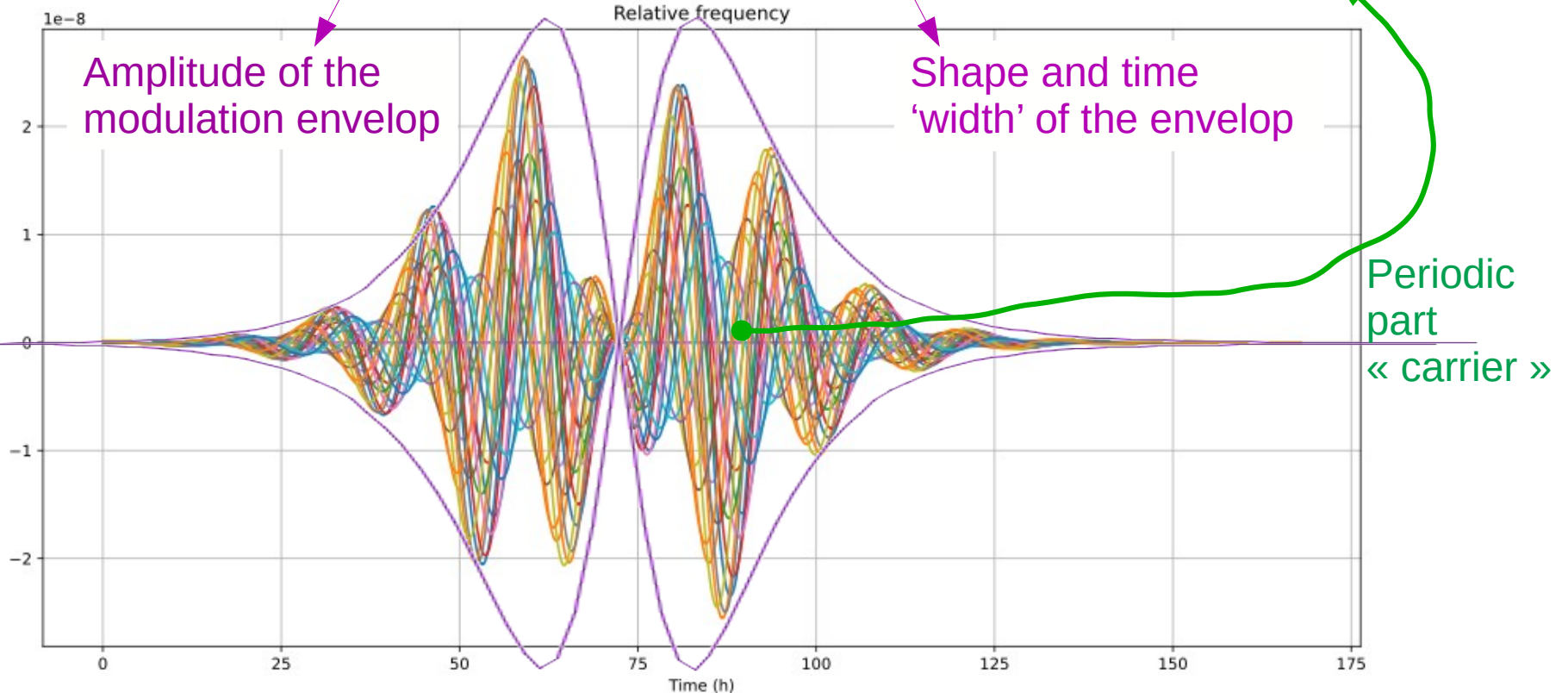
Statistical SNR threshold ρ_{thres}



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Theoretical modelisation for large transients

$$\Delta S^{ab}(t) = \tilde{A} \left[\frac{\Lambda v_{gal}}{\mathcal{T}}, \theta, \phi \right] \frac{\tanh \left[\frac{v_{\perp}}{d} (t - t_0) \right]}{\cosh^2 \left[\frac{v_{\perp}}{d} (t - t_0) \right]} \cos \left[\omega_{sat} t' + \tilde{\Phi}(\theta, \phi) \right]_{t_i}^t$$



$$\Delta s^{ab}(t) = \frac{3}{2} \frac{R_{\text{sat}}}{\omega_{\text{sat}}} \underbrace{\Gamma_{\text{eff}} \rho_{\text{DM}} v_{\text{gal}}}_{\text{Amplitude of the envelope (model part)}} \underbrace{\mathcal{T} \sqrt{C^2(\theta, \phi) + D^2(\theta, \phi)}}_{\text{Amplitude of the envelope (event \& satellite configuration part)}} \underbrace{\frac{\tanh\left[\frac{v_{\perp}}{d}(t - t_0)\right]}{\cosh^2\left[\frac{v_{\perp}}{d}(t - t_0)\right]}}_{\text{Shape \& time width of the envelope}} \cos \left[\omega_{\text{sat}} t' + \arctan \left(-\frac{C(\theta, \phi)}{D(\theta, \phi)} \right) \right]_{t_i}^t$$

Amplitude of the envelope
(model part)

$$\Gamma_{\text{eff}} \times \mathcal{T}$$

Degeneracy

Amplitude of the envelope
(event & satellite configuration part)

$$\theta, \phi$$

$$\Omega_a, u_a, i_a$$

Shape & time width of the envelope

$$\frac{v_{\perp}}{d} \quad t$$

Degeneracy

For large transient, the amplitude of the signal is independent of the size d of the transient

Identifying a given event to parameters space only imply two variables : the amplitude of the envelop and its width

How to construct the envelop in a statistically viable way from the stacking of satellites pairs ?

